NIVI5 GEOSPATIAL



LA County LIDAR PROCESSING REPORT

Project ID: 300076 Work Unit: 300610

Prepared for:



2024

Prepared by: NIV5 GEOSPATIAL

Submitted:September17,2024

National Map Help Desk: tnm_help@usgs.gov

Contents

1. Summary / Scope	1
1.1. Summary	1
1.2. Scope	1
1.3. Coverage	1
1.4. Duration	1
1.5. Issues	1
2. Planning / Equipment	4
2.1. Flight Planning	4
2.2. Lidar Sensor	4
2.3. Aircraft	6
2.4. Time Period	7
3. Processing Summary	
3.1. Flight Logs	8
3.2. Lidar Processing	
3.3. LAS Classification Scheme	
3.4. Classified LAS Processing	
3.5. Hydro-Flattened Breakline Processing	
3.6. Hydro-Flattened Raster DEM Processing	
3.7. Intensity Image Processing	
3.8. Maximum Surface Height Raster Processing	
3.9. Swath Separation Raster Processing	
3.10. Point Density	
4. Project Coverage Verification	
5. Accuracy Testing	
5.1. Calibration Control Point Testing	
5.2. Point Cloud Testing	
5.3. Digital Elevation Model (DEM) Testing	
6. Geometric Accuracy	
6.1. Horizontal Accuracy	24
6.2. Relative Vertical Accuracy (Interswath Accuracy)	25
6.3. Intraswath Precision (Smooth Surface Precision)	
Project Report Appendices	xxvii
Appendix A	xxviii
Flight Logs	xxviii
Appendix B	xxix
SBET and POSPac Reports	xxix

List of Figures

Figure 1. Work Unit Boundary	3
Figure 2. Riegl VQ1560iiS Lidar Sensor	5
Figure 3. NV5 Geospatial's Aircraft	6
Figure 4. Example of Swath Separation Raster	. 13
Figure 5. First Return Point Density	. 15
Figure 6. Ground First Return Point Density	. 16
Figure 7. Lidar Tile Layout	. 17
Figure 8. Lidar Coverage	. 18
Figure 9. Calibration Control Point Locations	. 21
Figure 10. QC Checkpoint Locations - NVA	. 22
Figure 11. QC Checkpoint Locations - VVA	. 23

List of Tables

Table 1. Originally Planned Lidar Specifications	1
Table 2. Lidar System Specifications	5
Table 3. LAS Classifications	10
Table 4. NVA and VVA Accuracy Results	20
Table 5. Intraswath Accuracy Results	26

List of Appendices

Appendix A: Flight Logs Appendix B: SBET and POSPac Reports

1. Summary / Scope

1.1. Summary

This report contains a summary of the LA County, Work Unit 300610 lidar acquisition task order, issued by USGS under their Contract 140G0221D0012 on August 17, 2023. The task order yielded a work unit area covering 107 square miles over California at Quality Level 1. The intent of this document is only to provide specific validation information for the data acquisition/collection, processing, and production of deliverables completed as specified in the task order.

1.2. Scope

Aerial topographic lidar was acquired using state of the art technology along with the necessary surveyed ground control points (GCPs) and airborne GPS and inertial navigation systems. The aerial data collection was designed with the following specifications listed in Table 1 below.

Table 1. Originally Planned Lidar Specifications

Average Point Density	Flight Altitude (AGL)	Field of View	Minimum Side Overlap	RMSEz
8 pts / m2	2532 m	58.5°	55%	≤ 10 cm

1.3. Coverage

The work unit boundary covers 107 square miles over California. Work unit extents are shown in Figure 1.

1.4. Duration

Lidar data was acquired from November 20, 2023 to December 7, 2023 in 4 total lifts. See "Section: 2.4. Time Period" for more details.

1.5. Issues

No issues to report.

LA County Work Unit 300610 Projected Coordinate System: UTM Zone 10 Horizontal Datum: NAD83 (2011) Vertical Datum: NAVD88 (GEOID 18) Units: Meters		
Lidar Point Cloud	Classified Point Cloud in .LAZ 1.4 format	
Rasters	 0.5-meter Hydro-flattened Bare Earth Digital Elevation Model (DEM) in GeoTIFF format 0.5-meter Intensity images in GeoTIFF format 1-meter Maximum Surface Height Raster 1-meter Swath Seperation Images 	
Vectors	 Shapefiles (*.shp) Project Boundary Lidar Tile Index Flightline Swath Calibration and QC Checkpoints (NVA/VVA) Geodatabase (*.gdb) Continuous Hydro-flattened Breaklines Geopackage (*.gpkg) Calibration and QC Checkpoints (NVA/VVA) 	
Reports	 Reports in PDF format Focus on Delivery Focus on Accuracy Survey Report Processing Report 	
Metadata	 XML Files (*.xml) Breaklines Classified Point Cloud DEM Intensity Imagery DSM 	





Figure 1. Work Unit Boundary

Page 3 of 26

2. Planning / Equipment

2.1. Flight Planning

Flight planning was based on the unique project requirements and characteristics of the project site. The basis of planning included: required accuracies, type of development, amount / type of vegetation within project area, required data posting, and potential altitude restrictions for flights in project vicinity.

Detailed project flight planning calculations were performed for the project using RiPARAMETER planning software.

2.2. Lidar Sensor

NV5 Geospatial utilized Riegl VQ1560iiS lidar sensors (Figure 2), serial number(s) 3546, 4040, 4046, 3061, , for data acquisition.

The Riegl 1560iiS system (System ID: R15S) is a dual channel waveform processing airborne scanning system with a pulse repetition rate of 4MHz, or up to 3900m AGL at a pulse repetition rate of 540kHz. This improved range increases the system's productivity. The system also utilizes a Multi-Pulse in the Air option (MPIA) and an integrated IMU/GNSS unit.

A brief summary of the aerial acquisition parameters for the project are shown in the lidar System Specifications in Table 2.

		Riegl VQ1560iiS
Terrain and	Flying Height	2532 m
Aircraft Scanner	Recommended Ground Speed	145 kts
	Field of View	58.5°
Scanner	Scan Rate Setting Used	204 lps
Lecor	Laser Pulse Rate Used	1,534 kHz
Laser	Multi Pulse in Air Mode	Yes
C	Full Swath Width	2837 m
Coverage	Line Spacing	0.707 m
Point Spacing	Average Nominal Point Spacing	0.35 m
and Density	Average Point Density	8 pts / m²

Table 2. Lidar System Specifications

Figure 2. Riegl VQ1560iiS Lidar Sensor



2.3. Aircraft

All flights for the project were accomplished through the use of customized aircraft. Plane type and tail numbers are listed below.

Lidar Collection Planes

• Cessna Caravan (single-turboprop), Tail Number(s): N604MD, N840JA, N22TE, N208JA

These aircraft provided an ideal, stable aerial base for lidar acquisition. These aerial platforms have relatively fast cruise speeds, which are beneficial for project mobilization / demobilization while maintaining relatively slow stall speeds, proving ideal for collection of high-density, consistent data posting using a state-of-the-art lidar system. NV5 Geospatial's operating aircraft can be seen in Figure 3 below.

Figure 3. NV5 Geospatial's Aircraft



2.4. Time Period

Project specific flights were conducted between November 20, 2023 to December 7. Four aircraft lifts were completed. Accomplished lifts are listed below.

Lift	Start UTC	End UTC
11102023A (SN3546,N604MD)	11/10/2023 8:13:55 PM	11/10/2023 11:12:17 PM
12012023A (SN3061,N208JA)	12/02/2023 2:59:52 AM	12/02/2023 5:05:35 AM
12022023A (SN3061,N208JA)	12/03/2023 3:55:12 AM	12/03/2023 5:19:52 AM
12072023A (SN3061,N208JA)	12/07/2023 9:41:41 PM	12/07/2023 10:43:57 PM

3. Processing Summary

3.1. Flight Logs

Flight logs were completed by Lidar sensor technicians for each mission during acquisition. These logs depict a variety of information, including:

- Job / Project #
- Flight Date / Lift Number
- FOV (Field of View)
- Scan Rate (HZ)
- Pulse Rate Frequency (Hz)
- Ground Speed
- Altitude
- Base Station
- PDOP avoidance times
- Flight Line #
- Flight Line Start and Stop Times
- Flight Line Altitude (AMSL)
- Heading
- Speed
- Returns
- Crab

Notes: (Visibility, winds, ride, weather, temperature, dew point, pressure, etc). Project specific flight logs for each sortie are available in Appendix A.

3.2. Lidar Processing

Applanix + POSPac software was used for post-processing of airborne GPS and inertial data (IMU), which is critical to the positioning and orientation of the lidar sensor during all flights. Applanix POSPac combines aircraft raw trajectory data with stationary GPS base station data yielding a "Smoothed Best Estimate Trajectory" (SBET) necessary for additional post processing software to develop the resulting geo-referenced point cloud from the lidar missions.

During the sensor trajectory processing (combining GPS & IMU datasets) certain statistical graphs and tables are generated within the Applanix POSPac processing environment which are commonly used as indicators of processing stability and accuracy. This data for analysis include: max horizontal / vertical GPS variance, separation plot, altitude plot, PDOP plot, base station baseline length, processing mode, number of satellite vehicles, and mission trajectory. Each sensor is initially factory calibrated. Further adjustment is performed on each sensor by periodically flying boresight locations and using this data to update boresight values used in data processing. Various proprietary tools and methodologies are used during this process. Once all data has been processed with updated boresight values, FL to FL match is performed by using strip align and other proprietary tools/processes.

Point clouds in flightline swath format were created using the RiPROCESS software. The generated point cloud is the mathematical three dimensional composite of all returns from all laser pulses as determined from the aerial mission. Each flightline swath point cloud was calibrated using Strip Align software that corrects systematic geometric errors and improves the relative and absolute accuracy of the flightline swath point cloud swaths were imported into GeoCue distributive processing software and the imported data was then tiled so further processing could take place in TerraScan software. Using TerraScan, the vertical accuracy of the surveyed ground control was tested and any vertical bias was removed from the data. TerraScan and TerraModeler software packages were then used for automated data classification and manual cleanup. The data were manually reviewed and any remaining artifacts removed using functionality provided by TerraScan and TerraModeler.

DEMs and Intensity Images are then generated using proprietary software. In the bare earth surface model, above-ground features are excluded from the data set. Global Mapper is used as a final check of the bare earth dataset.

Software	Version
Applanix + POSPac	8.9
RIPROCESS	1.9.3.6
Microstation Connect	10.16.02.34
GeoCue	2020.1.22.3
Global Mapper	19.1;20.1
TerraModeler	23.013
TerraScan	23.025
TerraMatch	23.013
StripAlign	2.21

Finally, proprietary software is used to perform statistical analysis of the LAS files.

3.3. LAS Classification Scheme

The classification classes are determined by Lidar Base Specifications 2023 Rev. A and are an industry standard for the classification of lidar point clouds. All data starts the process as Class 1 (Unclassified), and then through automated classification routines, the classifications are determined using TerraScan macro processing.

The classes used in the dataset are as follows and have the following descriptions:

	Classification Name	Description
1	Processed, but Unclassified	Laser returns that are not included in the bare earth class, or any other project classification
2	Bare earth	Laser returns that are determined to be bare earth using automated and manual cleaning algorithms
7	Low Noise	Laser returns that are often associated with scattering from reflective surfaces, or artificial points below the bare earth surface
9	Water	Laser returns that are found inside of hydro features
17	Bridge Deck	Laser returns falling on bridge decks
18	High Noise	Laser returns that are often associated with birds or artificial points above the bare earth surface
20	Ignored Ground	Bare earth points that fall within the given threshold of a collected hydro feature.

Table 3. LAS Classifications

3.4. Classified LAS Processing

The bare earth surface is then manually reviewed to ensure correct classification on the Class 2 (Ground) points. After the bare- earth surface is finalized; it is then used to generate all hydro-breaklines through heads-up digitization.

All ground (ASPRS Class 2) lidar data inside of the Lake Pond and Double Line Drain hydro flattening breaklines were then classified to water (ASPRS Class 9) using proprietary tools. A buffer of 1.5 feet/0.5 meter was also used around each hydro flattened feature to classify these ground (ASPRS Class 2) points to Ignored ground (ASPRS Class 20). All Lake Pond Island and Double Line Drain Island features were checked to ensure that the ground (ASPRS Class 2) points were reclassified to the correct classification after the automated classification was completed.

Any noise that was identified either through manual review or automated routines was classified to the appropriate class (ASPRS Class 7 and/or ASPRS Class 18) followed by flagging with the withheld bit.

All data was manually reviewed and any remaining artifacts removed using functionality provided by TerraScan and TerraModeler. Global Mapper is used as a final check of the bare earth dataset. GeoCue was then used to create the deliverable industry-standard LAS files for all point cloud data. NV5 Geospatial's proprietary software was used to perform final statistical analysis of the classes in the LAS files, on a per tile level to verify final classification metrics and full LAS header information.

3.5. Hydro-Flattened Breakline Processing

Using heads-up digitization, all Lake-Ponds, Double Line Drains, and Islands are manually collected that are within the project size specification. This includes Lake-Ponds greater than 2 acres in size, Double Line Drains with greater than a 100 foot nominal width, and Islands greater than 1 acre in size within a collected hydro feature. Lidar intensity imagery and bare-earth surface models are used to ensure appropriate and complete collection of these features.

Elevation values are assigned to all collected hydro features via NV5 Geospatial's proprietary software. This software sets Lake-Ponds to an appropriate, single elevation to allow for the generation of hydro-flattened digital elevation models (DEM). Double Line Drain elevations are assigned based on lidar elevations and surrounding terrain feature to ensure all breaklines match the lidar within acceptable tolerances. Some deviation is expected between breakline and lidar elevations due to monotonicity, connectivity, and flattening rules that are enforced on the breaklines. Once complete, horizontal placement, and vertical variances are reviewed, all breaklines are evaluated for topological consistency and data integrity using a combination of proprietary tools and manual review of hydro-flattened DEMs.

Breaklines are combined into one seamless shapefile, clipped to the project boundary, and imported into an Esri file geodatabase for delivery.

3.6. Hydro-Flattened Raster DEM Processing

Hydro-Flattened DEMs (topographic) represent a lidar-derived product illustrating the grounded terrain and associated breaklines (as described above) in raster form. NV5 Geospatial's proprietary software was used to take all input sources (bare earth lidar points, bridge and hydro breaklines, etc.) and create a Triangulated Irregular Network (TIN) on a tile-by-tile basis. Data extending past the tile edge is incorporated in this process so that proper triangulation can occur. From the TIN, linear interpolation is used to calculate the cell values for the raster product. The raster product is then clipped back to the tile edge so that no overlapping cells remain across the project area. A 32-bit floating point GeoTIFF DEM was generated for each tile with a pixel size of 0.5-meter. NV5 Geospatial's proprietary software was used to ensure a seamless dataset. NV5 Geospatial ensures there are no void or no-data values (-999999) in each derived DEM. This is achieved by using propriety software checking all cell values that fall within the project boundary. NV5 Geospatial uses a proprietary tool called FOCUS on Delivery to check all formatting requirements of the DEMs against what is required before final delivery.

3.7. Intensity Image Processing

Intensity images represent reflectivity values collected by the lidar sensor during acquisition. Proprietary software generates intensity images using first returns and excluding those flagged with a withheld bit. Intensity images are linearly scaled to a value range specific to the project area to standardize the images and reduce differences between individual tiles. Appropriate horizontal projection information as well as applicable header values are written during product generation.

3.8. Maximum Surface Height Raster Processing

Maximum Surface Height rasters (topographic) represent a lidar-derived product illustrating natural and builtup features. NV5 Geospatial's proprietary software was used to take all classified lidar points, excluding those flagged with a withheld bit, and create a raster on a tile-by-tile basis. Data extending past the tile edge is incorporated in this process so that proper gridding can occur. The raster is created by laying a 1-meter DEM cell size over the area and assigning the values to cells by using the maximum lidar point that intersects that grid cell. The raster product is then clipped back to the tile edge so that no overlapping cells remain across the project area. A 32-bit floating point GeoTIFF was then generated for each tile with a pixel size of 1-meter. There is no interpolation type being used in creating the raster product. NV5 Geospatial's proprietary software was used to write appropriate horizontal and vertical projection information as well as applicable header values into the file during product generation. Each maximum surface height raster is reviewed in Global Mapper to check for any anomalies and to ensure a seamless dataset. NV5 Geospatial uses a proprietary tool called FOCUS on Delivery to check all formatting requirements of the DEMs against what is required before final delivery

3.9. Swath Separation Raster Processing

Swath Separation Images are rasters that represent the interswath alignment between flight lines and provide a qualitative evaluation of the positional quality of the point cloud. NV5 Geospatial proprietary software generated 1-meter raster images in GeoTIFF format using last returns, excluding points flagged with the withheld bit, and using a point-in-cell algorithm. Images are generated with a 75% intensity opacity and (4) absolute 8-cm intervals, see below for interval coloring. Intensity images are linearly scaled to a value range specific to the project area to standardize the images and reduce differences between individual tiles. Appropriate horizontal projection information as well as applicable header values are written to the file during product generation. NV5 Geospatial uses a proprietary tool called FOCUS on Delivery to check all formatting requirements of the images against what is required before final delivery.



Figure 4. Example of Swath Separation Raster

0-8cm
8-16cm
16-24cm
>24cm

Page 13 of 26

3.10. Point Density

The acquisition parameters were designed to acquire an average first-return density of 8 points/m2. First return density describes the density of pulses emitted from the laser that return at least one echo to the system. Multiple returns greater than 1 from a single pulse were not considered in first return density analysis. Some types of surfaces (e.g., breaks in terrain, water, and steep slopes) may have returned fewer pulses than originally emitted by the laser. First returns typically reflect off the highest feature on the landscape within the footprint of the pulse. In forested or urban areas, the highest feature could be a tree, building or power line, while in areas of unobstructed ground, the first return will be the only echo and represents the bare earth surface.

The density of ground-classified lidar returns was also analyzed for this project. Terrain character, land cover, and ground surface reflectivity all influenced the density of ground surface returns. In vegetated areas, fewer pulses may penetrate the canopy, resulting in lower ground density.

The average first-return density of lidar data for the project was 12.79 points/m2 while the average ground classified density was 9.30 points/m2. The statistical and spatial distributions of first return densities and classified ground return densities per 100 m x 100 m cell are portrayed in Figures 5 and 6.





Figure 5. First Return Point Density

Page 15 of 26





Figure 6. Ground First Return Point Density

Page 16 of 26





Figure 7. Lidar Tile Layout

Page 17 of 26

4. Project Coverage Verification

A proprietary tool (FOCUS on Flight) produces grid-based polygons of each flightline, depicting exactly where lidar points exist. These swath polygons are reviewed against the project boundary to verify adequate project coverage. Please refer to Figure 8.



LA County Work Unit 300610 Lidar Coverage



5. Accuracy Testing

5.1. Calibration Control Point Testing

Figure 9 shows the location of each bare earth calibration point for the project area. TerraScan was used to perform a quality assurance check using the lidar bare earth calibration points. The results of the surface calibration are not an independent assessment of the accuracy of these project deliverables, but the statistical results do provide additional feedback as to the overall quality of the elevation surface.

5.2. Point Cloud Testing

The project specifications require that only Non-Vegetated Vertical Accuracy (NVA) be computed for raw lidar point cloud swath files. The required accuracy (ACCz) is: 19.6 cm at a 95% confidence level, derived according to NSSDA, i.e., based on RMSE of 10 cm in the "bare earth" and "urban" land cover classes. The NVA was tested with 1 checkpoint located in bare earth and urban (non-vegetated) areas. These check points were not used in the calibration or post processing of the lidar point cloud data. The checkpoints were distributed throughout the project area and were surveyed using GPS techniques. See survey report for additional survey methodologies.

Elevations from the unclassified lidar surface were measured for the x,y location of each check point. Elevations interpolated from the lidar surface were then compared to the elevation values of the surveyed control points. AccuracyZ has been tested to meet 19.6 cm or better Non-Vegetated Vertical Accuracy at 95% confidence level using RMSE(z) x 1.9600 as defined by the National Standards for Spatial Data Accuracy (NSSDA); assessed and reported using National Digital Elevation Program (NDEP)/ASPRS Guidelines.

5.3. Digital Elevation Model (DEM) Testing

The project specifications require the accuracy (ACCz) of the derived DEM be calculated and reported in two ways:

- The required NVA is: 19.6 cm at a 95% confidence level, derived according to NSSDA, i.e., based on RMSE of 10 cm in the "bare earth" and "urban" land cover classes. This is a required accuracy. The NVA was tested with 1 checkpoint located in bare earth and urban (non-vegetated) areas. See Figure 10.
- 2. Vegetated Vertical Accuracy (VVA): VVA shall be reported for "brushlands/low trees" and "tall weeds/ crops" land cover classes. The target VVA is: 29.4 cm at the 95th percentile, derived according to ASPRS Guidelines, Vertical Accuracy Reporting for lidar Data, i.e., based on the 95th percentile error in all vegetated land cover classes combined. This is a target accuracy. The VVA was tested with 3 checkpoints located in tall weeds/crops and brushlands/low trees (vegetated) areas. The checkpoints were distributed throughout the project area. See Figure 11.

AccuracyZ has been tested to meet 19.6 cm or better Non-Vegetated Vertical Accuracy at 95% confidence level using RMSE(z) x 1.9600 as defined by the National Standards for Spatial Data Accuracy (NSSDA); assessed and reported using National Digital Elevation Program (NDEP)/ASRPS Guidelines.

A brief summary of results are listed below.

	Target	Measured	Point Count
Raw NVA	0.196 m	0.0719 m	1
NVA	0.196 m	0.0094 m	1
VVA	0.294 m	0.2919 m	3

Table 4. NVA and VVA Accuracy Results

WA Central Wildfire Calibration Points





Page 21 of 26

LA County NVA Points





Page 22 of 26

LA County VVA Points





Page 23 of 26

6. Geometric Accuracy

6.1. Horizontal Accuracy

Lidar horizontal accuracy is a function of Global Navigation Satellite System (GNSS) derived positional error, flying altitude, and INS derived attitude error. The obtained RMSE, value is multiplied by a conversion factor of 1.7308 to yield the horizontal component of the National Standards for Spatial Data Accuracy (NSSDA) reporting standard where a theoretical point will fall within the obtained radius 95% of the time. Based on a flying altitude of 2500 meters, an IMU error of 0.002 decimal degrees, and a GNSS positional error of 0.019 meters, this project was compiled to meet 0.27 meter horizontal accuracy at the 95% confidence level. A summary is shown below.

Horizontal Accuracy		
RMSE _r	0.52 ft	
	0.15 m	
ACC _r	0.89 ft	
	0.27 m	

6.2. Relative Vertical Accuracy (Interswath Accuracy)

Relative vertical accuracy refers to the internal consistency of the data set as a whole: the ability to place an object in the same location given multiple flight lines, GPS conditions, and aircraft attitudes. When the lidar system is well calibrated, the swath-to-swath vertical divergence is low (<0.10 meters). The relative vertical accuracy was computed by comparing the ground surface model of each individual flight line with its neighbors in overlapping regions. The average (mean) line to line relative vertical accuracy for the LA County project was 0.071 feet (0.022 meters). A summary is shown below.

Relative Vertical Accuracy				
Sample	480 flight line surfaces			
Average	0.071 ft			
Average	0.022 m			
Madian	0.074 ft			
Median	0.022 m			
DMCF	0.092 ft			
RMSE	0.028 m			
Standard Doviation (1g)	0.040 ft			
Standard Deviation (10)	0.012 m			
1.00-	0.078 ft			
1.960	0.024 m			



6.3. Intraswath Precision (Smooth Surface Precision)

Intraswath Precision (smooth surface precision) is the measure of reliability of the lidar point cloud elevations along a planar surface. This measurement is performed on hard surfaces against a single flightline. NV5 digitized several large parking lots as polygons across the project area. These polygons were then used to calculate precision on a single FL basis using the below formula:

Precision = Range – (Slope x Cellsize x 1.414)

Range – Is the difference between the highest and lowest lidar points in each cell
Slope – is the maximum slope of the cell to its 8 neighbors
Cellsize – is set to the ANPS, rounded up to the next integer, and then doubled

NV5 calculated the RMSDz to be 1.7 cm, minimum slope-corrected range to be 0 cm, and the maximum slope-corrected range to be 6.6 cm.



Table 5. Intraswath Accuracy Results

Project Report Appendices

The following section contains the appendices as listed in

the LA County Lidar Project Report.

Appendix A

Flight Logs

LA County B23 Lidar Project - Work Unit 300610 Pa

Page xxviii of xxix

September 17, 2024

Project	946623-R037786.00	LA_County_Lidar_2023		
Flightplan	LA_CountyV2_1560iiS_8ppms			
Mission Name	S2223061_20231201_F1	Mission Notes		
Mission Date	12/1/2023	Was able to collect the majority of the western Channel Islands within the tide window. We spent most		
Aircraft	N208JA	of the day in mx in KFAT.		
Pilot	Matthew Gill			
Co-Pilot				
Operator	Logan Melgosa			
Co-Operator				
Vendor	NV5 Geospatial			
Base Airport	KCMA			
Departure (Local Time)	6:50:00 PM			
Arrival (Local Time)	9:33:00 PM			

Line	Heading	Start Time	Stop Time	Speed (kt)	Notes
		(010)	(010)		
00241	SW	02:59:52	03:08:09	133.8	
00242	NW	03:09:31	03:09:38	151.3	
00242	E	03:12:11	03:21:13	142.4	
00243	E	03:22:37	03:22:41	233.9	
00243	SW	03:26:03	03:37:41	127.5	
00244	E	03:41:21	03:52:28	140.3	
00245	SW	03:56:21	04:07:54	135.6	
00246	E	04:11:29	04:22:58	139.1	
00247	SW	04:26:50	04:38:36	136.8	
00248	E	04:41:30	04:53:10	138.1	

Project	946623-R037786.00	LA_County_Lidar_2023	
Flightplan	LA_CountyV2_1560iiS_8ppms		
Mission Name	S2223061_20231202_F1	Mission Notes	
Mission Date	12/2/2023	Was able to collect the majority of the Channel Islands for this project, we had a bit of a late start on our	
Aircraft	N208JA	first lift due to the FBO being slow to fuel. We will be conducting a second lift over KLAX tonight.	
Pilot	Matthew Gill		
Co-Pilot			
Operator	Logan Melgosa		
Co-Operator			
Vendor	NV5 Geospatial		
Base Airport	КСМА		
Departure (Local Time)	7:30:00 PM		
Arrival (Local Time)	10:15:00 PM		

Heading	Start Time (UTC)	Stop Time (UTC)	Speed (kt)	Notes
SW	03:55:11	04:06:26	141.6	Refly 5-15 statute miles FEE due to clouds
E	04:10:28	04:19:20	145.1	Refly 5-14 statute miles FEE due to clouds
SW	04:22:34	04:28:37	143.2	Refly 5-14 statute miles FEE due to clouds
E	04:31:45	04:37:36	144.5	Refly 6-13 statute miles FEE due to clouds
SW	04:40:39	04:45:40	142.9	Refly 4-11 statute miles FEE due to clouds
E	04:48:53	04:53:30	145.0	Refly 5-8 statute miles FEE due to clouds
SW	04:56:52	05:00:43	144.1	
E	05:04:00	05:07:21	147.7	
SW	05:10:50	05:13:27	142.8	
E	05:23:57	05:33:00	145.9	
SW	05:39:15	05:51:55	142.9	
	Heading SW E SW E SW E SW E SW E SW	Heading Start Time (UTC) SW 03:55:11 E 04:10:28 SW 04:22:34 E 04:31:45 SW 04:40:39 E 04:48:53 SW 04:56:52 E 05:04:00 SW 05:10:50 E 05:23:57 SW 05:39:15	Start Time (UTC) Stop Time (UTC) SW 03:55:11 04:06:26 E 04:10:28 04:19:20 SW 03:55:11 04:06:26 E 04:10:28 04:19:20 SW 04:22:34 04:28:37 E 04:31:45 04:37:36 SW 04:40:39 04:45:40 E 04:48:53 04:53:30 SW 04:56:52 05:00:43 E 05:04:00 05:07:21 SW 05:10:50 05:13:27 E 05:23:57 05:33:00 SW 05:39:15 05:51:55	Heading Start Time (UTC) Stop Time (UTC) Speed (kt) SW 03:55:11 04:06:26 141.6 E 04:10:28 04:19:20 145.1 SW 04:22:34 04:28:37 143.2 E 04:31:45 04:37:36 144.5 SW 04:44:39 04:45:40 142.9 E 04:44:53 04:53:30 145.0 SW 04:45:52 05:00:43 144.1 E 05:04:00 05:07:21 147.7 SW 05:10:50 05:13:27 142.8 E 05:23:57 05:33:00 145.9 SW 05:39:15 05:51:55 142.9

Project	946623-R037786.00	LA_County_Lidar_2023			
Flightplan	LA_CountyV2_1560iiS_8ppms				
Mission Name	S2223061_20231207_F1	Mission Notes			
Mission Date	12/7/2023	Was able to collect requested reflies over the Channel Islands without incident.			
Aircraft	N208JA				
Pilot	Matthew Gill				
Co-Pilot					
Operator	Logan Melgosa				
Co-Operator					
Vendor	NV5 Geospatial				
Base Airport	KHHR				
Departure (Local Time)	12:53:00 PM				
Arrival (Local Time)	3:29:00 PM				

Line	Heading	Start Time (UTC)	Stop Time (UTC)	Speed (kt)	Notes
00271	SW	21:36:09	21:39:56	135.8	
00249	SW	21:41:40	21:53:33	134.0	
00250	E	21:56:57	22:05:36	148.9	
00251	SW	22:08:51	22:15:32	129.5	
00252	E	22:18:04	22:23:47	147.9	
00253	SW	22:26:17	22:31:38	133.9	
00254	E	22:33:51	22:38:23	147.6	

Project	946623-R037786.00	LA_County_Lidar_2023
Flightplan	LA_CountyV2_1560iiS_8ppm	IS

Mission Name	S2223546_20231110_F2	Mission Notes
Mission Date	11/10/2023	Acquired data under high cirrus with smooth air.
Aircraft	N604MD	
Pilot	Grey Chandler	
Co-Pilot		
Operator	James Schoone	
Co-Operator		
Vendor	NV5 Geospatial	
Base Airport	KSBA	
Departure (Local Time)	11:51:00 AM	
Arrival (Local Time)	3:26:00 PM	

Line	Heading	Start Time (UTC)	Stop Time (UTC)	Speed (kt)	Notes
00241	SW	20:13:54	20:22:08	134.9	
00242	E	20:25:58	20:34:55	143.4	
00243	SW	20:38:28	20:49:28	135.1	
00244	E	20:51:41	21:02:31	143.7	
00245	SW	21:04:29	21:15:59	136.4	
00246	E	21:17:48	21:28:54	143.9	
00247	SW	21:30:52	21:42:42	136.1	
00248	E	21:44:31	21:55:44	143.4	
00249	SW	21:57:47	22:09:35	134.8	
00250	E	22:12:56	22:21:50	144.5	
00251	SW	22:24:07	22:30:12	142.0	
00252	E	22:31:59	22:37:51	144.5	
00253	SW	22:40:04	22:44:59	146.3	
00254	E	22:46:34	22:51:00	151.2	
00255	SW	22:52:40	22:56:21	150.6	
00256	E	22:57:52	23:01:11	148.6	
00257	SW	23:02:58	23:05:30	147.6	
	Ν	23:07:53	23:12:17	160.1	

Appendix B

SBET and POSPac Reports

Page xxix of xxix

September 17, 2024
General Information

Mission Information

Project name	11102023A_3546
Processing date	2024-01-12 20:36:12
Mission date	2023-11-10 19:43:57
Mission duration	03:18:25.309
Processing mode	IN-Fusion PP-RTX

Rover Hardware Information

Product	POS AV 610 VER6 HW2.5-12
Serial number	S/N9922
IMU type	57
Receiver type	BD982
Antenna type	AV59

Project File List

Rover Data Files

File name	File type
231110_194338_INS-GPS_1.raw	POS Data

Input Files

File Name	File Type
Ephm3140.23g	GLONASS Broadcast Ephemeris
Ephm3140.23n	GPS Broadcast Ephemeris
IGS00PSFIN_20233130000_01D_15M_0RB.SP3	GPS Precise Ephemeris
IGS00PSFIN_20233140000_01D_15M_ORB.SP3	GPS Precise Ephemeris
IGS00PSFIN_20233150000_01D_15M_ORB.SP3	GPS Precise Ephemeris

Output Files

Filename	File type	
sbet_11102023A_3546.out	SBET Trajectory File	

Rover Data Summary

First raw data file	231110_194338_INS-GPS_1.raw			
Last raw data file	231110_194338_INS-GPS_1.raw			
Start GPS week	2287			
Start time	503018.305 (11/10/2023 7:43:38 PM)			
End time	516546.638 (11/10/2023 11:29:06 PM)			
Start of fine alignment	504641.496 (11/10/2023 8:10:41 PM)			
Available subsystems	Primary GNSS, Gimbal, IMU			
POS Event Input	None			
Correction data	None			
IMU Installation Lever Arms & Mounting Angles				
Gimbal to IMU lever arm (m)	-0.034	-0.010	-0.374	
Gimbal to IMU mounting angles (deg)	0.000	0.000	0.000	
Gimbal to Primary GNSS lever arm (m)	0.205	0.058	-1.282	
Gimbal to Primary GNSS lever arm std dev (m)	-1.000			
Aircraft to Reference mounting angles (deg)	0.000	0.000	0.000	

Rover Data QC

Raw IMU Import QC Summary

IMU data input file	imu_Mission 1.dat
IMU data check log file	imudt_11102023A_3546.log
IMU Records Processed	2705046
Termination Status	Normal
IMU Anomalies	0

Primary Observables & Satellite Data

GPS/GLONASS L1 Satellite Lock/Elevation



^{- 0-10} deg - 10-15 deg - 15-20 deg - 20-30 deg - 30+ deg - No ephem + Cycle slip

GPS L1 SNR



GLONASS L1 SNR





GPS/GLONASS L2 Satellite Lock/Elevation

GPS L2 SNR



GLONASS L2 SNR



BEIDOU Satellite Lock/Elevation



BEIDOU SNR



GALILEO Satellite Lock/Elevation



GALILEO SNR





Smoothed Trajectory Information

Top View

Altitude



Roll/Pitch



Heading





North/East Velocity

Down Velocity



Total Speed



Ground Speed





Body Acceleration

Total Body Acceleration



Body Angular Rate





Forward Processed Trajectory Information

Top View

Altitude



Roll/Pitch



Heading





North/East Velocity

Down Velocity



Total Speed



Ground Speed





Body Acceleration

1.25 1.2 1.15 1.1 1.05 1 0.95 0.9 0.85 0.8 0.75 0.7 0.65 0.6 0.55 0.5 0.45 0.4 0.35 0.3 0.25 0.2 0.15 0.1 0.05 510,000 Time (sec) 506,000 508,000 512,000 514,000 516,000

Total Body Acceleration

Body Angular Rate



GNSS QC

GNSS QC Statistics

Statistics	Min	Max	Mean
Baseline length (km)	0.00	0.00	
Number of GPS SV	5	10	8
Number of GLONASS SV	3	6	5
Number of QZSS SV	0	0	0
Number of BEIDOU SV	0	0	0
Number of GALILEO SV	3	8	6
Total number of SV	12	22	19
PDOP	1.06	2.46	1.22
QC Solution Gaps	0.00	0.00	
Solution Type	Fixed	Float	No solution
Epoch (sec)	13477.00	0.00	0.00
Percentage	100.00	0.00	0.00

Num SVs in solution



- Number of GPS - Number of GLONASS - Number of QZSS - Number of BEIDOU - Number of GALILEO - Total Number

Forward/Reverse Separation



PDOP



Estimated Position Accuracy



GPS Residuals



GLONASS Residuals



GALILEO Residuals



GNSS-Inertial Processor Configuration

Processing mode	IN-Fusion PP-RTX			
Stabilized mount	True			
Processing start time	504640.691 (11/10/2023 8:10:40 PM)			
Processing end time	516546.000 (11/10/2023 11:29:06 PM)			
Initial attitude source	Primary GNSS Track			
IMU Sensor Context	Processing with Onboard IMU			
Gimbal to IMU lever arm (m)	-0.034	-0.010	-0.374	
Gimbal to IMU mounting angles (deg)	0.000	0.000	0.000	
Gimbal to Primary GNSS lever arm (m)	0.093	0.062	-1.287	
Gimbal to Primary GNSS lever arm std dev (m)	0.030	0.030	0.030	
Aircraft to Reference mounting angles (deg)	0.000	0.000	0.000	

Calibrated Installation Parameters

Reference-Primary GNSS Lever Arm (m)

X Reference-Primary GNSS Lever Arm (m)



Y Reference-Primary GNSS Lever Arm (m)



Z Reference-Primary GNSS Lever Arm (m)



Reference-Primary GNSS Lever Arm Figure of Merit



IN-Fusion QC

Forward Processed Estimated Errors, Reference Frame

Accelerometer Bias (micro-g)





X Accelerometer Bias (micro-g)

Y Accelerometer Bias (micro-g)



Z Accelerometer Bias (micro-g)











X Accelerometer Scale Error (ppm)







Z Accelerometer Scale Error (ppm)

Gyro Bias (deg/h)





X Gyro Bias (deg/h)









Gyro Scale Error (ppm)





X Gyro Scale Error (ppm)



Y Gyro Scale Error (ppm)

Z Gyro Scale Error (ppm)

506,000

508,000



510,000 Time (sec)

512,000

514,000

516,000


Smoothed Performance Metrics

Position Error RMS (m)

Velocity Error RMS (m/s)





Roll/Pitch Error RMS (arc-min)

0.78 0.76 0.74 0.72 0.7 0.68 0.66 0.64 0.62 0.6 0.58 0.56 0.54 0.52 0.5 0.48 0.46 0.44 0.42 0.4 0.38 0.36 0.34 0.32 0.3 0.28 0.26 0.24 510,000 Time (sec) 506,000 508,000 512,000 514,000 516,000

Heading Error RMS (arc-min)

Forward Processed Performance Metrics

Position Error RMS (m)





Velocity Error RMS (m/s)

- North Velocity Error RMS (m/s) - East Velocity Error RMS (m/s) - Down Velocity Error RMS (m/s)

Roll/Pitch Error RMS (arc-min)



Heading Error RMS (arc-min)



Forward Processed Solution Status

Processing Mode



Number of Satellites



Baseline Length



General Information

Mission Information

Project name	12022023A_3061
Processing date	2023-12-05 17:53:15
Mission date	2023-12-03 03:19:22
Mission duration	03:01:58.559
Processing mode	IN-Fusion PP-RTX

Rover Hardware Information

Product	POS AV 610 VER6 HW2.5-12
Serial number	S/N8902
IMU type	57
Receiver type	BD982
Antenna type	AV59

Project File List

Rover Data Files

File name	File type
231203_031433_INS-GPS_1.raw	POS Data

Input Files

File Name	File Type
Ephm3370.23g	GLONASS Broadcast Ephemeris
Ephm3370.23n	GPS Broadcast Ephemeris

Output Files

Filename	File type
sbet_12022023A_3061.out	SBET Trajectory File

Rover Data Summary

First raw data file	231203_031433_INS-GPS_1.raw			
Last raw data file	231203_031433_INS-GPS_1.raw			
Start GPS week	2291			
Start time	11943.072 (12/3/2023 3:19:03 AM)			
End time	22861.631 (12/3/2023 6:21:01 AM)			
Start of fine alignment	12147.328 (12/3/2023 3:22:27 AM)			
Available subsystems	Primary GNSS, Gimbal, IMU			
POS Event Input	None			
Correction data	None			
IMU Installation Lever Arms & Mounting Angles				
Gimbal to IMU lever arm (m)	0.000	0.000	0.000	
Gimbal to IMU mounting angles (deg)	0.000	0.000	0.000	
Gimbal to Primary GNSS lever arm (m)	0.000	0.000	0.000	
Gimbal to Primary GNSS lever arm std dev (m)	-1.000			
Aircraft to Reference mounting angles (deg)	0.000	0.000	0.000	

Rover Data QC

Raw IMU Import QC Summary

IMU data input file	imu_Mission 1.dat
IMU data check log file	imudt_12022023A_3061.log
IMU Records Processed	2183267
Termination Status	Normal
IMU Anomalies	0

Primary Observables & Satellite Data

GPS/GLONASS L1 Satellite Lock/Elevation



- 0-10 deg - 10-15 deg - 15-20 deg - 20-30 deg - 30+ deg - No ephem + Cycle slip

GPS L1 SNR



GLONASS L1 SNR



GPS/GLONASS L2 Satellite Lock/Elevation



GPS L2 SNR



GLONASS L2 SNR



- GLONASS 16 LZ SNR (dB/HZ) - GLONASS 17 LZ SNR (dB/HZ) - GLONASS 18 LZ SNR (dB/HZ) - GLONASS 19 LZ SNR (dB/HZ) - GLONASS 20 LZ SNR (dB/HZ) - GLONASS 21 LZ SNR (dB/HZ)



BEIDOU Satellite Lock/Elevation



BEIDOU SNR

GALILEO Satellite Lock/Elevation



GALILEO SNR





Smoothed Trajectory Information

Top View

Altitude



Roll/Pitch



Heading





North/East Velocity

Down Velocity



Total Speed



Ground Speed





Body Acceleration



Total Body Acceleration

Body Angular Rate





Forward Processed Trajectory Information Top View

Altitude



Roll/Pitch



Heading





North/East Velocity

Down Velocity



Total Speed



Ground Speed





Body Acceleration



Total Body Acceleration

Body Angular Rate



GNSS QC

GNSS QC Statistics

Statistics	Min	Max	Mean
Baseline length (km)	0.00	0.00	
Number of GPS SV	2	9	8
Number of GLONASS SV	0	6	5
Number of QZSS SV	0	0	0
Number of BEIDOU SV	0	0	0
Number of GALILEO SV	5	8	6
Total number of SV	10	22	19
PDOP	1.02	2.30	1.21
QC Solution Gaps	2.00	2.00	
Solution Type	Fixed	Float	No solution
Epoch (sec)	10858.00	0.00	2.00
Percentage	99.98	0.00	0.02

Num SVs in solution



- Number of GPS - Number of GLONASS - Number of QZSS - Number of BEIDOU - Number of GALILEO - Total Number



Forward/Reverse Separation









GPS Residuals



GLONASS Residuals



0.065 0.06 0.055 0.05 0.045 0.04 0.035 0.03 0.025 0.02 0.015 0.01 0.005 0 -0.005 -0.01 -0.015 -0.02 -0.025 -0.03 -0.035 -0.04 -0.045 -0.05 17,000 18,0 Time (sec) 13,000 14,000 15,000 16,000 18,000 19,000 20,000 21,000 22,000 GALILEO 02 Residual (m) — GALILEO 08 Residual (m) GALILEO 15 Residual (m) — GALILEO 25 Residual (m) – GALILEO 10 Residual (m) – GALILEO 11 Residual (m) – GALILEO 27 Residual (m) – GALILEO 30 Residual (m) GALILEO 34 Residual (m) — GALILEO 36 Residual (m)

GALILEO Residuals

GNSS-Inertial Processor Configuration

Processing mode	IN-Fusion PP-RTX		
Stabilized mount	True		
Processing start time	11944.000 (12/3/2023 3:19:04 AM)		
Processing end time	22861.000 (12/3/2023 6:21:01 AM)		
Initial attitude source	Primary GNSS Track		
IMU Sensor Context	Processing with Onboard IMU		
Gimbal to IMU lever arm (m)	-0.034	-0.010	-0.374
Gimbal to IMU mounting angles (deg)	0.000	0.000	0.000
Gimbal to Primary GNSS lever arm (m)	0.056	0.055	-1.294
Gimbal to Primary GNSS lever arm std dev (m)	0.030	0.030	0.030
Aircraft to Reference mounting angles (deg)	0.000	0.000	0.000

Calibrated Installation Parameters

Reference-Primary GNSS Lever Arm (m)

X Reference-Primary GNSS Lever Arm (m)



Y Reference-Primary GNSS Lever Arm (m)



Z Reference-Primary GNSS Lever Arm (m)



Reference-Primary GNSS Lever Arm Figure of Merit



IN-Fusion QC

Forward Processed Estimated Errors, Reference Frame Accelerometer Bias (micro-g)



— x accelerometer bias (micro-g) — y accelerometer bias (micro-g) — z accelerometer bias (micro-g)



X Accelerometer Bias (micro-g)

Y Accelerometer Bias (micro-g)



Z Accelerometer Bias (micro-g)


Accelerometer Scale Error (ppm)



X Accelerometer Scale Error (ppm)



Y Accelerometer Scale Error (ppm)



Z Accelerometer Scale Error (ppm)



Gyro Bias (deg/h)



X Gyro Bias (deg/h)





Y Gyro Bias (deg/h)

Z Gyro Bias (deg/h)



Gyro Scale Error (ppm)





X Gyro Scale Error (ppm)



Y Gyro Scale Error (ppm)



Z Gyro Scale Error (ppm)



Smoothed Performance Metrics

Velocity Error RMS (m/s)



37



Roll/Pitch Error RMS (arc-min)



Heading Error RMS (arc-min)



Forward Processed Performance Metrics

Position Error RMS (m)



Velocity Error RMS (m/s)



Roll/Pitch Error RMS (arc-min)

Heading Error RMS (arc-min)



Forward Processed Solution Status

Processing Mode



Number of Satellites



Baseline Length



General Information

Mission Information

Project name	12072023A_3061
Processing date	2023-12-12 18:42:09
Mission date	2023-12-07 20:47:52
Mission duration	02:44:09.309
Processing mode	IN-Fusion PP-RTX

Rover Hardware Information

Product	POS AV 610 VER6 HW2.5-12
Serial number	S/N8902
IMU type	57
Receiver type	BD982
Antenna type	AV59

Project File List

Rover Data Files

File name	File type
231207_204734_INS-GPS_1.raw	POS Data

Input Files

File Name	File Type
Ephm3410.23g	GLONASS Broadcast Ephemeris
Ephm3410.23n	GPS Broadcast Ephemeris

Output Files

Filename	File type
sbet_12072023A_3061.out	SBET Trajectory File

Rover Data Summary

First raw data file	231207_204734_INS-GPS_1.raw			
Last raw data file	231207_204734_INS-GPS_1.raw			
Start GPS week	2291			
Start time	420454.080 (12/7/2023 8:47:34 PM)			
End time	430303.389 (12/7/2023 11:31:43 PM)			
Start of fine alignment	420746.051 (12/7/2023 8:52:26 PM)			
Available subsystems	Primary GNSS, Gimbal, IMU			
POS Event Input	None			
Correction data	None			
IMU Installation Lever Arms & Mounting Angles				
Gimbal to IMU lever arm (m)	0.000	0.000	0.000	
Gimbal to IMU mounting angles (deg)	0.000	0.000	0.000	
Gimbal to Primary GNSS lever arm (m)	0.000	0.000	0.000	
Gimbal to Primary GNSS lever arm std dev (m)	-1.000			
Aircraft to Reference mounting angles (deg)	0.000 0.000 0.000			

Rover Data QC

Raw IMU Import QC Summary

IMU data input file	imu_Mission 1.dat
IMU data check log file	imudt_12072023A_3061.log
IMU Records Processed	1969506
Termination Status	Normal
IMU Anomalies	0

Primary Observables & Satellite Data

GPS/GLONASS L1 Satellite Lock/Elevation



- 0-10 deg - 10-15 deg - 15-20 deg - 20-30 deg - 30+ deg - No ephem + Cycle slip



GPS L1 SNR

GLONASS L1 SNR



GPS/GLONASS L2 Satellite Lock/Elevation





GPS L2 SNR

GLONASS L2 SNR



— GLONASS 11 L2 SNR (dB/Hz) — GLONASS 12 L2 SNR (dB/Hz) — GLONASS 20 L2 SNR (dB/ — GLONASS 21 L2 SNR (dB/Hz) — GLONASS 22 L2 SNR (dB/Hz)



BEIDOU Satellite Lock/Elevation



BEIDOU SNR

GALILEO Satellite Lock/Elevation



GALILEO SNR



Smoothed Trajectory Information

Top View



Altitude



Roll/Pitch



Heading



11



North/East Velocity

Down Velocity



Total Speed



Ground Speed





Body Acceleration

Total Body Acceleration



Body Angular Rate







Forward Processed Trajectory Information

Top View

Altitude







Heading





North/East Velocity

Down Velocity



Total Speed



Ground Speed





Body Acceleration



Total Body Acceleration

Body Angular Rate



GNSS QC

GNSS QC Statistics

Statistics	Min	Max	Mean
Baseline length (km)	0.00	0.00	
Number of GPS SV	2	11	8
Number of GLONASS SV	0	8	5
Number of QZSS SV	0	0	0
Number of BEIDOU SV	0	0	0
Number of GALILEO SV	4	6	5
Total number of SV	7	24	18
PDOP	1.04	2.99	1.24
QC Solution Gaps	2.00	4.00	
Solution Type	Fixed	Float	No solution
Epoch (sec)	9826.00	0.00	10.00
Percentage	99.90	0.00	0.10

Num SVs in solution



- Number of GPS - Number of GLONASS - Number of QZSS - Number of BEIDOU - Number of GALILEO - Total Number

Forward/Reverse Separation



PDOP





Estimated Position Accuracy



GPS Residuals
GLONASS Residuals



⁻ GLONASS 09 Residual (m) - GLONASS 10 Residual (m) - GLONASS 11 Residual (m) - GLONASS 12 Residual - GLONASS 20 Residual (m) - GLONASS 21 Residual (m) - GLONASS 22 Residual (m)



GALILEO Residuals

GNSS-Inertial Processor Configuration

Processing mode	IN-Fusion PP-RTX		
Stabilized mount	True		
Processing start time	420454.025 (12/7/2023 8:47:34 PM)		
Processing end time	430303.000 (12/7/2023 11:31:43 PM)		
Initial attitude source	Primary GNSS Track		
IMU Sensor Context	Processing with Onboard IMU		
Gimbal to IMU lever arm (m)	-0.034	-0.010	-0.374
Gimbal to IMU mounting angles (deg)	0.000	0.000	0.000
Gimbal to Primary GNSS lever arm (m)	0.056	0.055	-1.294
Gimbal to Primary GNSS lever arm std dev (m)	0.030	0.030	0.030
Aircraft to Reference mounting angles (deg)	0.000	0.000	0.000

Calibrated Installation Parameters

Reference-Primary GNSS Lever Arm (m)

X Reference-Primary GNSS Lever Arm (m)



Y Reference-Primary GNSS Lever Arm (m)



Z Reference-Primary GNSS Lever Arm (m)



Reference-Primary GNSS Lever Arm Figure of Merit



IN-Fusion QC

Forward Processed Estimated Errors, Reference Frame

Accelerometer Bias (micro-g)





X Accelerometer Bias (micro-g)

Y Accelerometer Bias (micro-g)



Z Accelerometer Bias (micro-g)



Accelerometer Scale Error (ppm)





X Accelerometer Scale Error (ppm)

Y Accelerometer Scale Error (ppm)



Z Accelerometer Scale Error (ppm)



Gyro Bias (deg/h)





X Gyro Bias (deg/h)



Y Gyro Bias (deg/h)



Z Gyro Bias (deg/h)

Gyro Scale Error (ppm)





X Gyro Scale Error (ppm)



Y Gyro Scale Error (ppm)

Z Gyro Scale Error (ppm)





Smoothed Performance Metrics

Velocity Error RMS (m/s)



Roll/Pitch Error RMS (arc-min)





Heading Error RMS (arc-min)



Forward Processed Performance Metrics

Position Error RMS (m)

Velocity Error RMS (m/s)







Heading Error RMS (arc-min)





Processing Mode



Number of Satellites



Baseline Length

